

AN OVERVIEW OF GPS AUGMENTATION SYSTEMS

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Abstract

The mature satellite-based navigation systems that are now available (e.g., GPS) have provided adequate positioning capability to users. However, their very success has been a driving force to increase accuracy, availability, reliability, and integrity requirements. As a result, several satellite augmentation systems have been or are in the process of being designed, developed, and/or tested in order to meet the ever-demanding requirements. This paper will provide a summary of the current possibilities to improve GPS performance, namely the impact of the GPS modernization program itself, augmentation with the satellite-based GLONASS, WAAS, MSAS, and EGNOS systems, and augmentation with on-board aiding; e.g., barometers and clocks. Performances are discussed as a function of user mask angle. The impact of combined GPS/GALILEO is briefly addressed.

WHY AUGMENTATION?

- ◆ Standalone GPS is not adequate for many applications in terms of [1]:
 - *Integrity* - the ability to protect the user from inaccurate information in a timely manner
 - *Accuracy* - the difference between measured and true positions of a vehicle at any given time
 - *Continuity* - the ability to complete an operation without triggering an alarm
 - *Availability* - the ability to be used by the user whenever it is needed

[1] - Loh, R., et al., (1995) "The U.S. Wide-Area Augmentation System (WAAS)", Navigation: Journal of The Institute of Navigation, Vol 42, No 3, Fall 1995, pp. 435-465.

“AUGMENTATION” OPTIONS

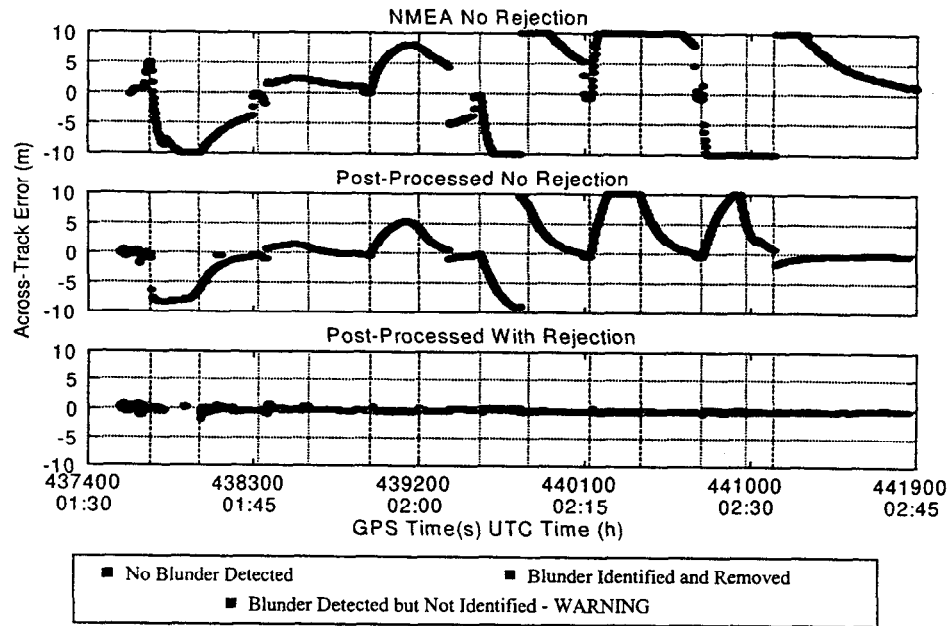
- ◆ Receiver algorithms (RAIM)
- ◆ Additional sensors
- ◆ Extra systems
 - GLONASS
 - GNSS2 - Galileo
- ◆ GPS Modernization
- ◆ Local Area Augmentation Systems (LAAS)
- ◆ Wide Area Augmentation Systems (WAAS)
 - EGNOS, US WAAS, MSAS

RECEIVER ALGORITHMS

- ◆ Receiver Autonomous Integrity Monitoring (RAIM)
- ◆ Fault Detection and Exclusion (FDE)
- ◆ Simple implementations can produce a significant reliability improvements [e.g., 2]
- ◆ Requires five or more satellites
 - Reduces/limits availability

[2] - Ryan, S., J. Stephen and G. Lachapelle, (1999) "Testing and Analysis of Reliability Measures for GNSS Receivers in the Marine Environment", Proceedings of the ION NTM-99, The Institute of Navigation, Alexandria, VA., pp. 505-514.

Example of RAIM/FDE



[2] - Ryan, S., J. Stephen and G. Lachapelle, (1999) "Testing and Analysis of Reliability Measures for GNSS Receivers in the Marine Environment", Proceedings of the ION NTM-99, The Institute of Navigation, Alexandria, VA., pp. 505-514.

ADDITIONAL ON-BOARD SENSORS

- ◆ Use of additional or complementary on-board sensors to monitor and/or augment GPS
 - Altimeter
 - Precise clock
 - Rate gyro
 - Compass
 - INS
- ◆ Vehicle Autonomous Integrity Monitoring (VAIM)
 - All on-board sensors contribute to navigation reliability

GLONASS

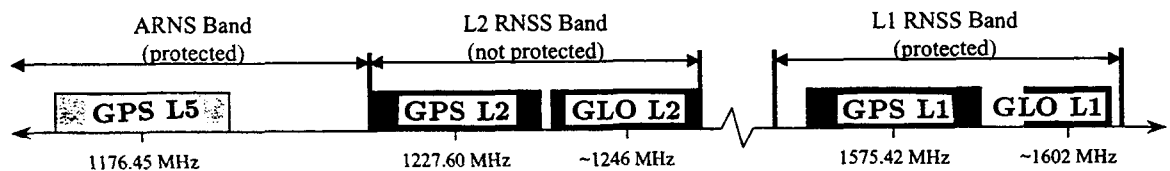
- ◆ Constellation of 10 operational satellites (as of November 30, 1999)
- ◆ Signal transmit on two frequencies
- ◆ No intentional degradation of ranging signal
- ◆ Large improvement in availability and reliability when combined with GPS
- ◆ Future of the system is uncertain

GNSS2 – GALILEO

- ◆ New satellite system conceived by the European Community (EC)
- ◆ Completion planned for 2008
- ◆ Constellation of 24+ satellites
 - Increased availability and reliability over GPS only
- ◆ Three to four carrier frequencies
 - Increased reliability

GPS MODERNIZATION

- ◆ Currently, L2 is not in a protected RF band
 - 3rd frequency needed for safety-of-life
- ◆ 2nd and 3rd civil frequencies confirmed
 - 1227.60 MHz (L2)
 - 1176.45 MHz (ARNS band), first launch 2005
- ◆ Higher power levels
- ◆ More robust code-modulation techniques



[3] - McDonald, K., (1999) "Opportunity Knocks: Will GPS Modernization Open Doors?", GPS World, Vol 10, No 9, September 1999, Advanstar Communications, pp. 36-46.

FAA SPECIFICATIONS

Requirement	Category I	Category II	Category III
Vertical Position Accuracy **	4.0 m	2.5 m	2.5 m
Integrity	4e-8 / approach	4e-8 / approach	1e-9 / approach
Time-to-Alert	6 seconds	2 seconds	2 seconds
Vertical Alarm Limit	10 m	5 m	5 m
Continuity	1e-5 / approach	1e-5 / approach	1e-7 / 30s

WAAS

LAAS

[4] - Federal Aviation Administration, (1999) "Local Area Augmentation System (LAAS) Update" Available at URL: <http://gps.faa.gov/Library/Documents/documents.htm#laas>

WIDE AREA AUGMENTATION SYSTEMS (WAAS)

Phases of Flight		Integrity	Availability	Accuracy
En Route	Oceanic	GPS with RAIM		
	Domestic	WAAS		
Approach & Landing	Non-Precision Approaches	WAAS		
	Category I Precision Approach	WAAS and LAAS		
	Category II/III Precision Approach	LAAS		
Surface	Ground Movement	LAAS		

- ◆ Major push from aviation community
- ◆ Designed to allow sole use of GPS for all phases of flight through Category I precision approach

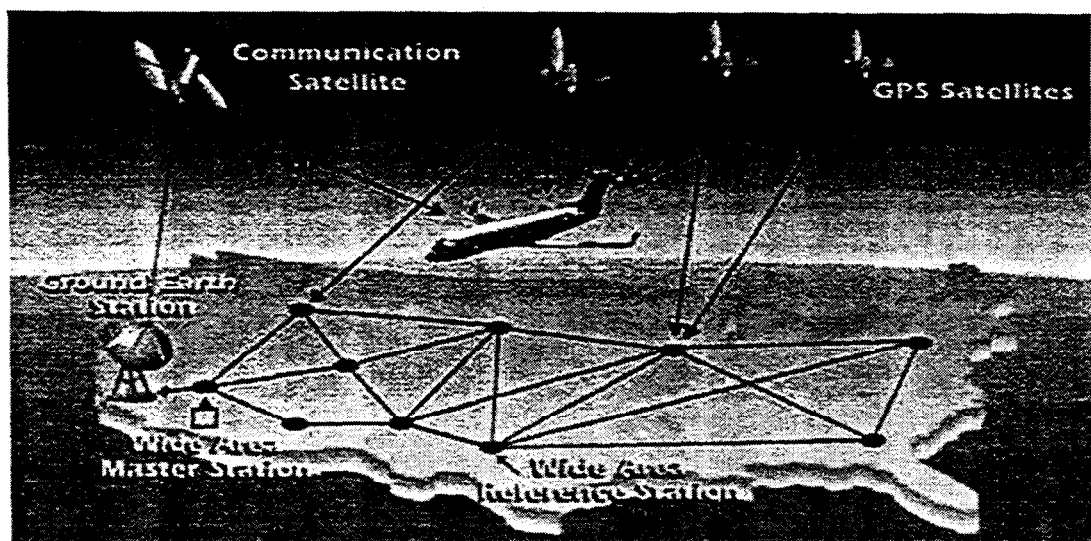
[5] - Hanlon, D. and K. Sandhoo, (1997) "FAA Satellite Navigation Program Overview", Proceedings of the ION Annual Meeting, The Institute of Navigation, Alexandria, VA., pp. 49-56.

- ◆ Three basic functions of a WAAS
 - Ranging
 - Provide additional ranging signals to improve availability, typically via geo-synchronous satellites
 - Integrity Channel
 - Provide transmission of GPS and integrity data to navigators
 - Wide Area Differential (WAD)
 - Provide differential correction data to users to improve accuracy
 - Satellite orbit and clock errors
 - Differential range corrections
 - Ionospheric grid computation

US WAAS

- ◆ Wide area Reference Station (WRS)
 - Collect and process data
- ◆ Wide area Master Station (WMS)
 - Compute all corrections to be received by users
- ◆ Ground Earth Station (GES)
 - Transmission to geo-synchronous satellites
- ◆ Communication Satellites (GEO)
 - Broadcast corrections and ranging signal

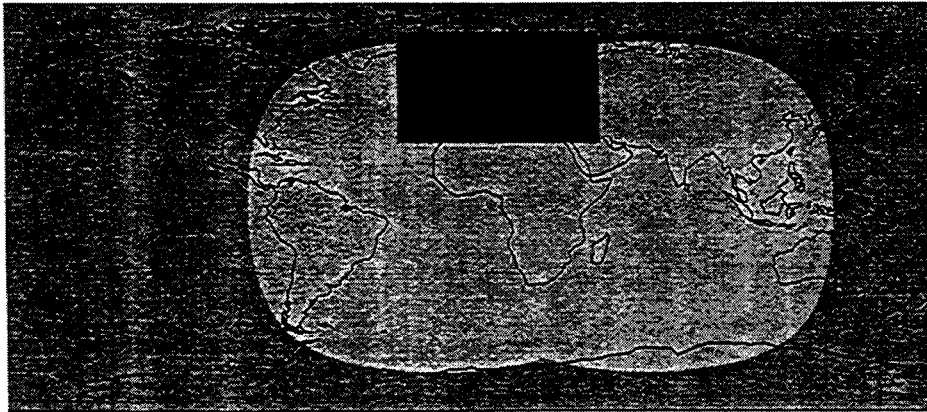
US WAAS Concept



Picture taken from URL - <http://www.iscns.com/GPSWSLS.HTM#WAAS>

EUROPEAN WAAS – EGNOS

- ◆ European Geostationary Navigation Overlay System (EGNOS)
- ◆ Similar to US WAAS but includes GLONASS satellite corrections as well



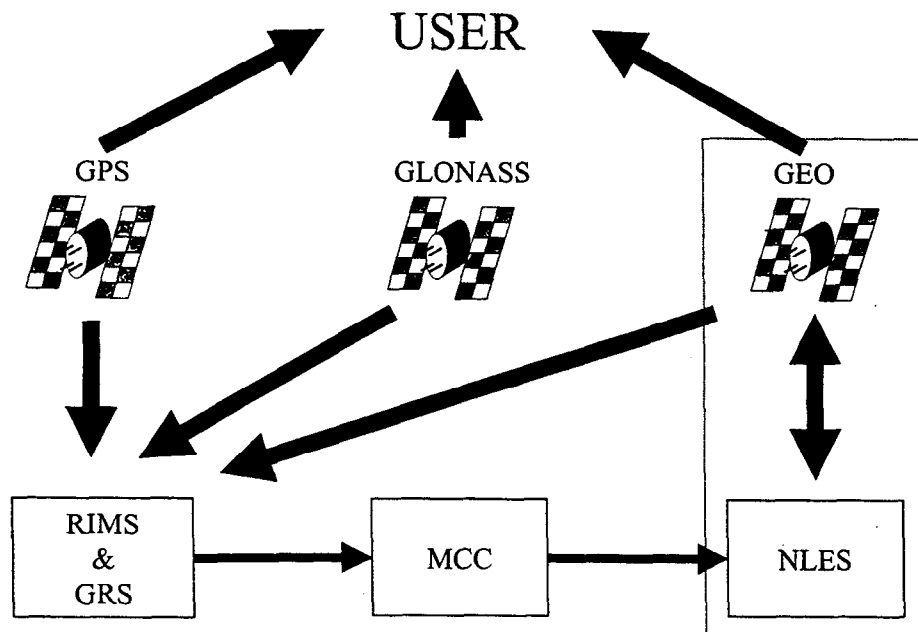
- EGNOS Service Volume
- ▨ INMARSAT Coverage

- ◆ **Ground Segment**
 - **Ranging and Integrity Monitoring Stations (RIMS)**
 - Collect range measurements and send them to the MCC
 - **Master Control Centre (MCC)**
 - Computation, distribution, validation, and transmission of data
 - Manage and control entire EGNOS system
 - **Geostationary Reference Station (GRS)**
 - Monitor geostationary satellites
 - Geostationary orbit determination
 - **Navigation Land Earth Station (NLES)**
 - Generate GPS-like signal centered on GPS L1 (1575.42 MHz) modulated with C/A code and navigation message (correction data)
 - Broadcast through geostationary satellites
 - Closed-loop control to maintain EGNOS system time

[6] - Loddo, et al., "EGONS, the European Regional Augmentation to GPS and GLONASS", The Proceedings of ION GPS-99, The Institute of Navigation, Alexandria, VA., pp. 1143-1150.

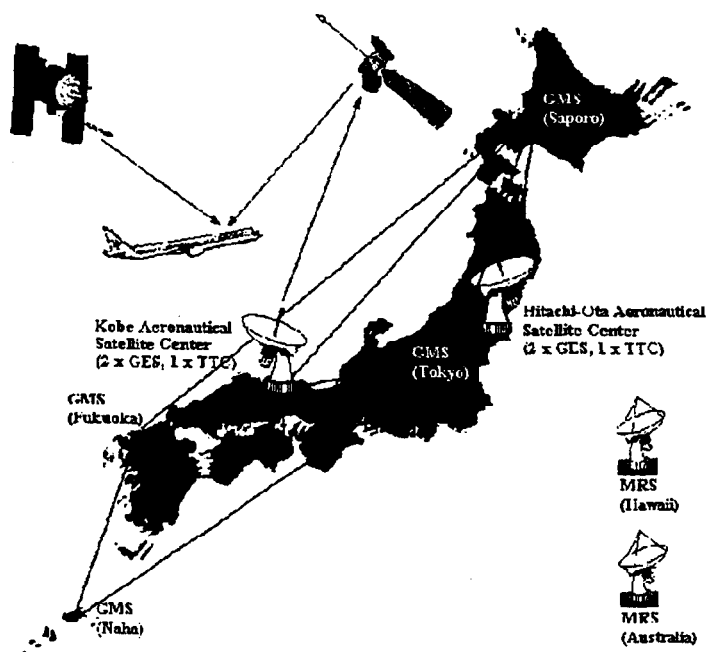
European WAAS - EGNOS

- ◆ **Space Segment**
 - GPS satellites
 - GLONASS satellites
 - INMARSAT III satellites for data transmission and ranging function (GEO)
- ◆ **User segment**
 - Signal in Space (SIS)
 - Receiver capable of receiving and decoding the GEO broadcast message



JAPANESE WAAS – MSAS

- ◆ MTSAT (Multi-Functional Transport SATellite) based Satellite Augmentation System (MSAS)
 - ◆ Similar to EGNOS system (GPS and GLONASS)
 - ◆ Limited geographical extent may lead to problems with orbit determination
 - Dynamic approach to orbit determination
 - Orbital relaxation approach
- ◆ Ground Segment
 - Ground Monitor Stations (GMS)
 - Collect range measurements and send them to the MCS
 - Monitor and Ranging Stations (MRS)
 - Receive GPS/MTSAT signals and collect range data
 - Master Control Stations (MCS)
 - Monitor and control system
 - Calculate MTSAT orbit, ionospheric delay, and correction data
 - Determine system integrity
 - Collect range data (GPS and MTSAT)
 - Send data to NES for uplink to MTSAT for broadcast
 - Navigation Earth Stations (NES)
 - Uplinks data from MCS to MTSAT for broadcast

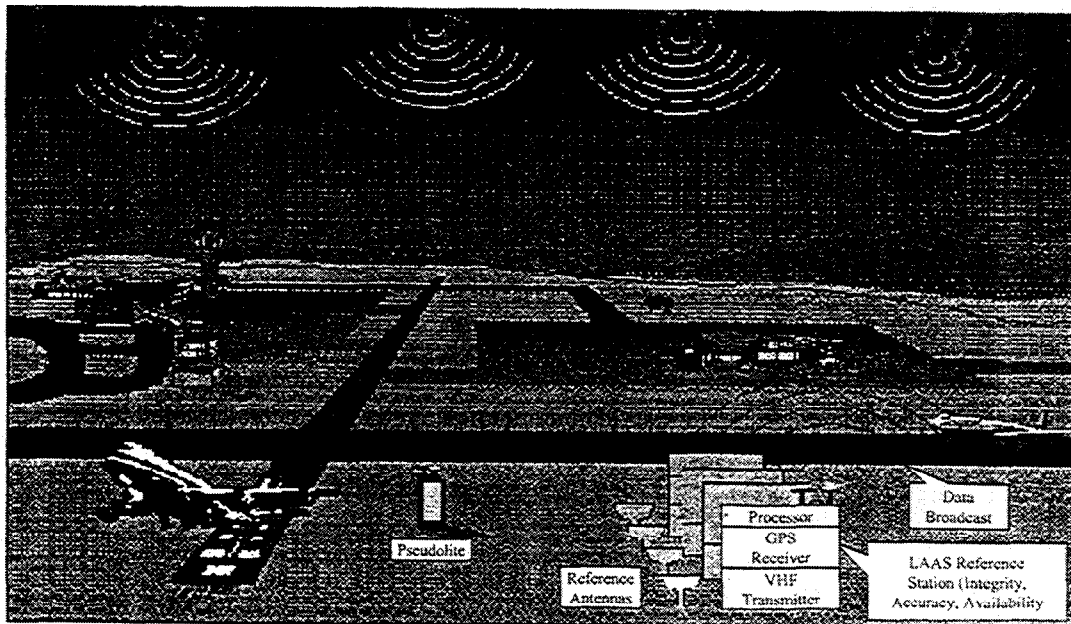


- ◆ Two Aeronautical Satellite Centers which include:
 - MRS's
 - 8 GMS's
 - 1 MCS's
- ◆ Launch of MTSAT satellite failed (November, 1999)
 - Rocket booster failure

LOCAL AREA AUGMENTATION SYSTEMS (LAAS)

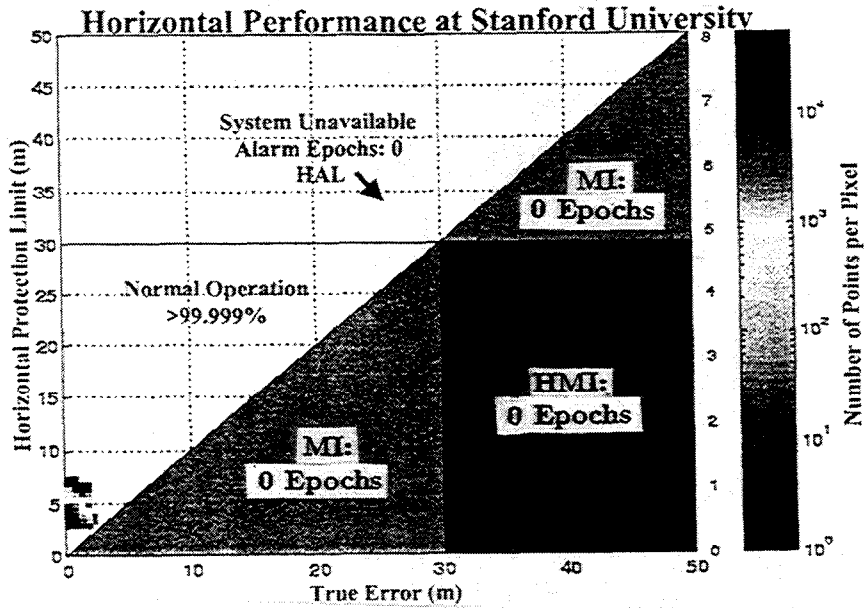
- ◆ FAA initiative to use GPS for all categories of precise landing, including CAT III
- ◆ Major differences from WAAS include
 - Limited range (~30 nm)
 - Limited number of base stations (~4)
 - Single differential correction to account for all errors
 - Smoothed-code or carrier-phase approaches are necessary
 - Ranging improvement through pseudolites

LAAS Concept

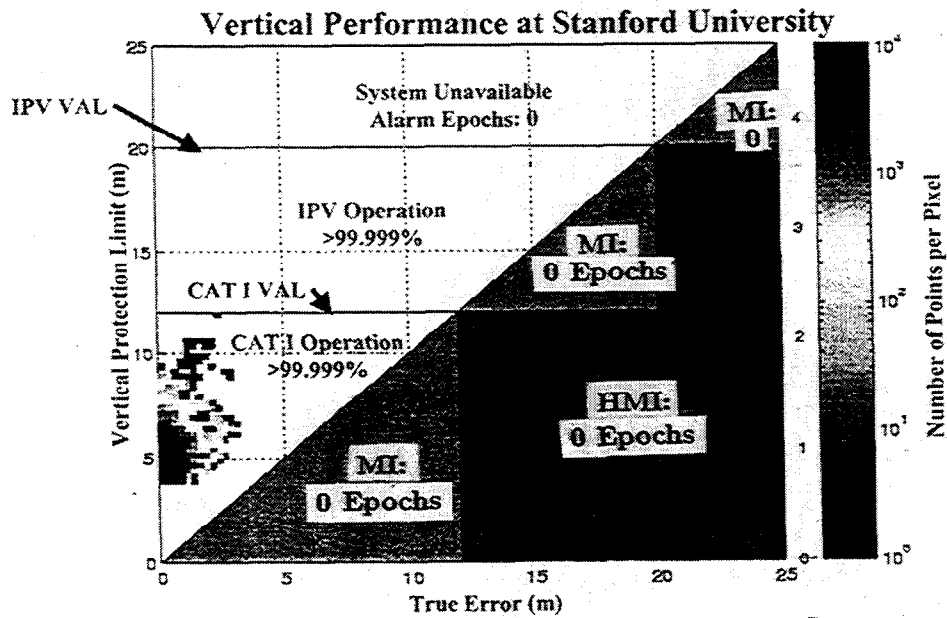


[4] - Federal Aviation Administration, (1999) "Local Area Augmentation System (LAAS) Update" Available at URL: <http://gps.faa.gov/Library/Documents/documents.htm#laas>

WAAS RESULTS



MI - Misleading Information
 HMI - Hazardously Misleading Information
 HAL - Horizontal Alarm Limit



MI - Misleading Information
 HMI - Hazardously Misleading Information
 IPV - Instrument Precision with Vertical Guidance
 VAL - Vertical Alarm Limit

Taken from URL - <http://www.stanford.edu/group/GPS/Projects/WAAS/metrics.html>

CONCLUSIONS

- ◆ SPS GPS is not robust enough for all applications
- ◆ Augmentation by WAAS, EGNOS, and MSAS will provide a true GNSS with high integrity, accuracy, and availability

Questions and Answers

MARC WEISS (NIST): I assume, when you showed the improvement with WAAS over the accuracy without WAAS, that was with SA turned on.

PATRICK FENTON (NovAtel): SA was on in both cases, yes.

WEISS: So, with SA turned off –

FENTON: WAAS not only helps with a clock, but is also an orbit computation. With SA off, you'd be probably sitting around 3 to 5 meters just with the orbit uncertainties where WAAS is also going to correct the orbit.

WEISS: So it's at a factor of two, with SA turned off, in improvement?

FENTON: Yes, I would guess that.

DAVID ALLAN (Allan's Time): How do the errors scale with the change in SA level? The current SA level is a peacetime level. Should that increase, which it could, how do the errors go with that level?

FENTON: Well, I think there's a specification for that in the Raytheon system. I might defer to the Raytheon folks next. But it wouldn't be linear because they don't actually broadcast a range rate. They broadcast SA corrections on something like a 6-second time basis, and you have to extrapolate through them. So it would increase linearly with increase of SA. I don't have a good number for that.